

On The Propagation Of Plane Waves At A Pure Shear Interface*

Wasiq Hussain[†]

Received 12 January 2007

Abstract

Propagation of elastic waves at a pure shear interface of two half-spaces with different strain-energy functions is studied. An interesting result of the existence of two interfacial waves (with no transmitted waves) in the upper half-space is found. The wave amplitudes are illustrated graphically highlighting the dependence on the finite pure shear deformation and the angle of incidence.

1 Introduction

In [1] Hussain and Ogden considered the effect of simple shear deformation on the reflection and transmission of homogeneous plane waves at the boundary between two half-spaces consist of the same material and corresponding to the same strain-energy functions (see also the related paper [2]). In the present paper the effect of *pure shear* on the reflection and transmission of plane waves at the boundary between two half-spaces corresponding to *different strain-energy* functions to consist of same incompressible isotropic elastic material is considered.

The required equations and notations are summarized in Section 2 and in the Subsections the propagation of plane harmonic waves is discussed with reference to the *slowness curves* appropriate for the two distinct classes of strain-energy functions.

For the *mixed case* of the strain-energy functions, the method for finding the amplitudes of the reflected, transmitted and interfacial waves is discussed in Section 3.1. For each angle of incidence a single reflected wave, with angle of reflection equal to the angle of incidence, is generated when a homogeneous plane (SV) wave is incident on the boundary from one half-space, and it is accompanied by an interfacial wave. It is shown that under a certain restriction on the state of deformation a transmitted (homogeneous plane SV) wave and an interfacial wave are generated for all angles of incidence. When this restriction does not hold there exists a critical angle such that (a) in $x_2 < 0$, above which, there is a reflected wave and an interfacial wave but below which there are two reflected waves (and no interfacial waves), and (b) in $x_2 > 0$,

*Mathematics Subject Classifications: 74Bxx,74Jxx.

[†]Department of Mathematics, School of Science and Engineering, Lahore University of Management Sciences (LUMS), Opposite Sector 'U', D.H.A., Lahore Cantt. 54792, Pakistan.

above which, there is again a transmitted and an interfacial wave but below which there are two interfacial waves (and no transmitted wave).

The theory in Section 3 is illustrated in Section 4 using graphical results to show the dependence of the amplitudes of the waves on the angle of incidence for representative values of the deformation parameters.

2 Basic Equations

Consider an incompressible isotropic elastic material subject to pure shear. Let $\lambda_1, \lambda_2, \lambda_3$ denote the principal stretches of the deformation. Then, the incompressibility condition is expressed as

$$\lambda_1 \lambda_2 \lambda_3 = 1.$$

If σ_i ($i = 1, 2, 3$) are the principal Cauchy stresses then the pure shear deformation confined to the (1-2)-plane is given by

$$\lambda_1 = \lambda \neq 1, \quad \lambda_2 = \lambda^{-1}, \quad \lambda_3 = 1 \quad \text{with } \sigma_1 \neq 0, \quad \sigma_2 = 0,$$

where a non-vanishing stress σ_3 is required to maintain $\lambda_3 = 1$.

The equation of motion in terms of a scalar function, $\psi(x_1, x_2, t)$ say, is given by

$$\alpha \psi_{,1111} + 2\beta \psi_{,1122} + \gamma \psi_{,2222} = \rho(\ddot{\psi}_{,11} + \ddot{\psi}_{,22}), \quad (1)$$

as in [1], where $,i$ denotes $\partial/\partial x_i$, $i \in \{1, 2\}$, ρ is the mass density, a superposed dot indicates the material time derivative and the constants α, β, γ are defined by

$$\alpha = \mathcal{A}_{01212}, \quad \gamma = \mathcal{A}_{02121}, \quad 2\beta = \mathcal{A}_{01111} + \mathcal{A}_{02222} - 2\mathcal{A}_{01122} - 2\mathcal{A}_{01221}. \quad (2)$$

In (2) \mathcal{A}_{0jilk} are the components of the fourth-order tensor \mathbf{A}_0 of instantaneous elastic moduli (see, for example, Ogden [3]).

The shear and normal components of the incremental nominal traction Σ_{21}, Σ_{22} on a plane $x_2 = \text{constant}$ are expressible in terms of ψ through

$$\begin{aligned} \Sigma_{21} &= \gamma \psi_{,22} - (\gamma - \sigma_2) \psi_{,11}, \\ -\Sigma_{22,1} &= (2\beta + \gamma - \sigma_2) \psi_{,112} + \gamma \psi_{,222} - \rho \ddot{\psi}_{,2}. \end{aligned} \quad (3)$$

2.1 Plane Waves

Consider time-harmonic homogeneous plane waves of the form

$$\psi = A \exp[ik(x_1 \cos \theta + x_2 \sin \theta - ct)], \quad (4)$$

where A is a constant, c (> 0) the wave speed, k (> 0) the wave number and $(\cos \theta, \sin \theta)$ the direction cosines of the direction of propagation of the wave in the (x_1, x_2) -plane. Substitution of (4) into (1) gives

$$\alpha \cos^4 \theta + 2\beta \sin^2 \theta \cos^2 \theta + \gamma \sin^4 \theta = \rho c^2. \quad (5)$$

Equation (5) is a relationship between the wave speed and the propagation direction in the (x_1, x_2) -plane and is called the propagation condition. The material constants are taken to satisfy the strong ellipticity inequalities

$$\alpha > 0, \quad \gamma > 0, \quad \beta > -\sqrt{\alpha\gamma}, \quad (6)$$

and it is clear from (5) that $\rho c^2 > 0$ if and only if (6) hold.

Similarly, from (1), for an inhomogeneous plane wave of the form

$$\psi = \hat{A} \exp[ik'(x_1 - imx_2 - c't)], \quad (7)$$

we obtain

$$\alpha - 2\beta m^2 + \gamma m^4 = \rho(1 - m^2)c'^2, \quad (8)$$

which relates the wave speed c' to the 'inhomogeneity factor' m . Note that the wave decays exponentially as $x_2 \rightarrow -\infty(+\infty)$ provided m has positive (negative) real part.

Consider two distinct cases corresponding to different strain-energy functions. For these either $2\beta = \alpha + \gamma$ or $2\beta \neq \alpha + \gamma$.

2.1.1 Case A: $2\beta = \alpha + \gamma$

For this case equations (5) and (8) reduce to

$$\alpha \cos^2 \theta + \gamma \sin^2 \theta = \rho c^2 \quad (9)$$

and

$$(m^2 - 1)(\alpha - \gamma m^2 - \rho c'^2) = 0 \quad (10)$$

respectively.

In terms of the *slowness vector* (s_1, s_2) defined by

$$(s_1, s_2) = (\cos \theta, \sin \theta)/c$$

equation (9) becomes the *slowness curve* (by using $\alpha/\gamma = \lambda^4$)

$$\lambda^4 s_1^2 + s_2^2 = \bar{\rho}, \quad (11)$$

in the (s_1, s_2) -space, where $\bar{\rho}$ is defined by

$$\bar{\rho} = \rho/\gamma. \quad (12)$$

By using the dimensionless notation (\bar{s}_1, \bar{s}_2) defined by

$$(\bar{s}_1, \bar{s}_2) \equiv (s_1, s_2)/\sqrt{\bar{\rho}}, \quad (13)$$

we can write (11) as

$$\lambda^4 \bar{s}_1^2 + \bar{s}_2^2 = 1. \quad (14)$$

2.1.2 Case B: $2\beta \neq \alpha + \gamma$

In this case we take the strain-energy function to satisfy $\beta = \sqrt{\alpha\gamma}$ which was used by Hussain and Ogden in [1]. Then (5) takes the form

$$[\sqrt{\alpha} \cos^2 \theta + \sqrt{\gamma} \sin^2 \theta]^2 = \rho c^2 \quad (15)$$

and (8) becomes

$$(\sqrt{\alpha} - \sqrt{\gamma} m^2)^2 = \rho(1 - m^2) c'^2. \quad (16)$$

The slowness curve corresponding to (15) is given by

$$[\lambda^2 \bar{s}_1^2 + \bar{s}_2^2]^2 = \bar{s}_1^2 + \bar{s}_2^2, \quad (17)$$

in dimensionless form with the notation (13) and $\bar{\rho}$ defined by (12). We now show graphically the dependence of the slowness curves on λ for both classes of strain-energy functions in (\bar{s}_1, \bar{s}_2) -space with reference to (14) and (17) (See Figs. 1-2).

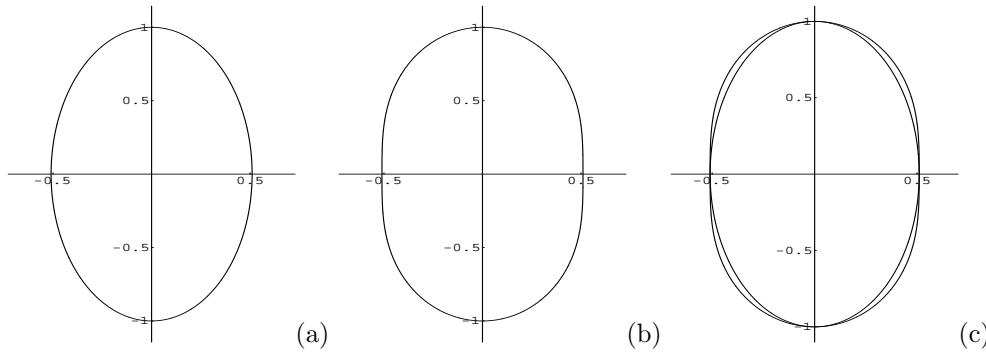


Figure 1: Slowness curves in (\bar{s}_1, \bar{s}_2) -space for $\lambda = 1.4$ with (a) $2\beta = \alpha + \gamma$, (b) $2\beta \neq \alpha + \gamma$, (c) the superposition of Figs. in (a) and (b).

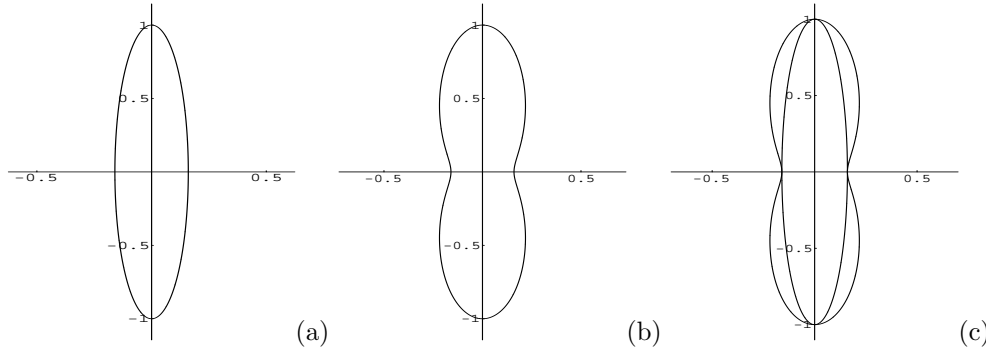


Figure 2: Slowness curves in (\bar{s}_1, \bar{s}_2) -space for $\lambda = 2.5$ with (a) $2\beta = \alpha + \gamma$, (b) $2\beta \neq \alpha + \gamma$, (c) the superposition of Figs. in (a) and (b).

3 Reflection and transmission at the interface

Consider the case in which a half-space defined by $x_2 < 0$ is joined to a half-space $x_2 > 0$. Both are subjected to pure shear deformation. The boundary conditions corresponding to continuous incremental displacement across the interface $x_2 = 0$ in terms of the scalar functions ψ and ψ^* , take the forms

$$\psi_{,1} = \psi_{,1}^*, \quad \psi_{,2} = \psi_{,2}^*, \quad (18)$$

respectively on $x_2 = 0$, where an asterisk signifies a quantity in $x_2 > 0$.

In terms of ψ , boundary conditions corresponding to continuous incremental traction, across the interface $x_2 = 0$ are recast through

$$\Sigma_{21} = \Sigma_{21}^*, \quad \Sigma_{22,1} = \Sigma_{22,1}^*,$$

on use of (3), in the forms

$$\begin{aligned} \psi_{,11} - \psi_{,22} &= \psi_{,11}^* - \psi_{,22}^*, \\ (2\beta + \gamma)(\psi_{,112} - \psi_{,112}^*) + \gamma(\psi_{,222} - \psi_{,222}^*) - \rho(\ddot{\psi}_{,2} - \ddot{\psi}_{,2}^*) &= 0. \end{aligned} \quad (19)$$

Because of the symmetry of slowness curves, with respect to the normal direction to the interface, the solution comprising the incident wave, a reflected wave and an interfacial wave in $x_2 < 0$ is written as

$$\begin{aligned} \psi &= A \exp[ik(x_1 \cos \theta + x_2 \sin \theta - ct)] + AR \exp[ik(x_1 \cos \theta - x_2 \sin \theta - ct)] \\ &\quad + AR' \exp[ik'(x_1 - imx_2 - c't)], \end{aligned} \quad (20)$$

where R is the reflection coefficient and R' measures the amplitude of the interfacial wave. The notations k' , m , c' are as used in (7) and m has positive real part.

Accordingly, in $x_2 > 0$, the wave solution may be written

$$\psi^* = AR^* \exp[ik^*(x_1 \cos \theta^* + x_2 \sin \theta^* - c^*t)] + AR^{*'} \exp[ik^{*'}(x_1 + im^*x_2 - c^{*'}t)], \quad (21)$$

comprising a transmitted and an interfacial wave, where R^* is the transmission coefficient and $R^{*'}$ is the analogue of R' for $x_2 > 0$. Note that the interfacial wave decays as $x_2 \rightarrow \infty$ provided m^* has positive real part.

Here Snell's law takes the form

$$\cos \theta / c = 1/c' = \cos \theta^* / c^* = 1/c^{*'}. \quad (22)$$

(22) states in particular, that the first components of the slowness vectors for each homogeneous plane wave interacting at the boundary $x_2 = 0$ are equal.

Thus, by reference to the slowness curves (superimposed) as exemplified in Fig. 1(c) and Fig. 2(c), the range of angles of incidence for which a transmitted wave exists can be identified. In Fig. 1(c), for example, if the *outer curve* corresponds to $x_2 < 0$ there is, for every angle of incidence (i.e. for every s_1 associated with the curve) a point on the inner curve (corresponding to $x_2 > 0$), and hence a transmitted wave.

In Fig. 2(c) on the other hand, there are values of s_1 on the slowness (outer) curve for $x_2 < 0$ for which there are no corresponding values on the inner slowness curve (corresponding to $x_2 > 0$), and therefore a range of angles of incidence for which no transmitted wave exists. This will be discussed further in Section 3.1.

We now examine here the case in which $2\beta \neq \alpha + \gamma$ ($x_2 < 0$), $2\beta = \alpha + \gamma$ ($x_2 > 0$).

3.1 $2\beta \neq \alpha + \gamma$ ($x_2 < 0$), $2\beta = \alpha + \gamma$ ($x_2 > 0$)

In this case we see from (16), after using Snell's law $\cos \theta/c = 1/c'$, we have

$$(m^2 + t^2)[m^2(1 + t^2) - t^2 + \lambda^2(\lambda^2 - 2)] = 0, \quad (23)$$

where $t = \tan \theta$. Note that t should be distinguished from the time variable t used earlier. $m = +it$ and $m = -it$ are the solutions of (23) corresponding to the incident and reflected waves respectively. The other solutions are

$$m = \pm \sqrt{1 - (\lambda^2 - 1)^2 / (1 + t^2)}. \quad (24)$$

If $\lambda \leq \sqrt{2}$ the m is real for all θ and the positive solution of (24) corresponds to an interfacial wave in $x_2 < 0$. If $\lambda > \sqrt{2}$ then there is a critical value of θ , θ_c say, for which $m = 0$ and this is given by

$$t_c^2 = \lambda^2(\lambda^2 - 2), \quad (25)$$

where the notation $t_c = \tan \theta_c$ is used. It follows that m is real for $\theta_c \leq \theta \leq \pi/2$. For $\theta_c < \theta \leq \pi/2$ there is a reflected wave accompanied by an interfacial wave and for $\theta = \theta_c$ the interfacial wave becomes a plane shear (body) wave propagating parallel to the boundary in $x_2 < 0$ (grazing reflection). When $0 < \theta < \theta_c$ the interfacial wave is replaced by a second reflected wave with angle of reflection, θ' say, obtained from (24) by replacing m by $-i \tan \theta'$ to give

$$t'^2 = \{\lambda^2(\lambda^2 - 2) - t^2\} / (1 + t^2),$$

where $t' = \tan \theta'$.

In $x_2 > 0$, from the counterpart of (10) we see that $m^* = \pm 1$, which yields an interfacial wave in the half-space $x_2 > 0$ for $m^* = +1$. The zeros of the other quadratic factor correspond to $m^* = i \tan \theta^*$ and $m^* = -i \tan \theta^*$, where $m^* = i \tan \theta^*$ corresponds to a transmitted wave provided $\tan \theta^*$ is real and positive. The value of $\tan \theta^*$ is obtained by using the propagation condition (15) and the counterpart of (9) together with Snell's law (22). This gives

$$t^{*2} = t^2 \{t^2 + \lambda^2(2 - \lambda^2)\} / (1 + t^2), \quad (26)$$

and the notation $t^* = \tan \theta^*$ has been introduced.

(26) shows that t^{*2} is positive when $t^2 > \lambda^2(\lambda^2 - 2) = t_c^2$ and negative when $t^2 < t_c^2$. Therefore there is a transmitted wave accompanied by an interfacial wave when $\theta \in (\theta_c, \pi/2]$ and for $\theta \in (0, \theta_c)$ the transmitted wave is replaced by a second interfacial

wave. To best of the knowledge of author, the *phenomenon of two interfacial waves* (in the upper half-space), *never appeared in linear elasticity*. In $x_2 > 0$ at $\theta = \theta_c$ there will be grazing transmission.

The coefficients R , R' , R^* and $R^{*'}$ are determined by using the boundary conditions (18), and (19), with the second equation in (19) taking the form

$$(2\lambda^2 + 1)\psi_{,112} - (\lambda^4 + 2)\psi_{,112}^* + \psi_{,222} - \psi_{,222}^* - \bar{\rho}(\ddot{\psi}_{,2} - \ddot{\psi}_{,2}^*) = 0 \quad (27)$$

in this case, where $\bar{\rho}$ is given by (12).

Substitution of ψ and ψ^* from (20) and (21)(with $m^* = 1$) in (18), first equation in (19) and (27) leads to

$$\begin{aligned} 1 + R + R' &= R^* + R^{*'}, \\ t(1 - R) - imR' &= t^*R^* + iR^{*'}, \\ (1 + R)(t^2 - 1) - (1 + m^2)R' &= (t^{*2} - 1)R^* - 2R^{*'}, \\ (R - 1)it(m^2 + 1) + R'm(t^2 - 1) &= -R^*it^*(2 - m^2t^2 + t^{*2}) - R^{*'}(t^{*2} - 1). \end{aligned} \quad (28)$$

In the latter equation use has been made of (23) and (26) in order to simplify the coefficients. In these equations, for given t , m is obtained from (24) and t^* from (26). The values of R , R' , R^* , and $R^{*'}$ are obtained from the solution of (28). The resulting expressions are not given here, but using Mathematica [4], graphical results showing the dependence of $|R|$, $|R'|$, $|R^*|$, and $|R^{*'}$ on θ and different values of λ are given in Section 4.

4 Numerical Results

Graphical results for a selection of values of λ are given in Figs. (3-6). With reference to the slowness curves (superimposed) in Fig.1(c), which is the relevant one for the Figs. (3-6)(a), it can be seen that there is one reflected wave, one transmitted wave and two interfacial waves for each possible angle of incidence when $2\beta \neq \alpha + \gamma$ ($x_2 < 0$), $2\beta = \alpha + \gamma$ ($x_2 > 0$). At $\theta = \pi/2$, incident wave is transmitted fully, as $|R| = |R'| = |R^{*'}| = 0$. At grazing incidence ($\theta = 0$), there are no interfacial waves.

In respect of Figs. (3-6)(c & d), from the discussion of slowness curves in Fig.2(c) it is apparent that there are two interfacial waves in $x_2 > 0$ and two reflected waves in $x_2 < 0$ for $0 \leq \theta \leq \theta_c$. A reflected wave with an interfacial wave (in $x_2 < 0$) and a transmitted wave along with an interfacial wave (in $x_2 > 0$) exist for $\theta_c \leq \theta \leq \pi/2$, where the critical angle θ_c is given by (25). For $\lambda = 1.5, 1.6, 1.73$, and 2.5 the value of $\theta_c = 0.512, 0.962, 1.25$, and 1.53 respectively. Note the continuity of graphs at $\theta = \theta_c$. Each of $|R|$, $|R^*|$, and $|R^{*'}$ vanish at one value of θ when $\lambda > \sqrt{2}$.

In Figs. (4-6) the change in the vertical scales must be noted. In addition, an interesting conclusion is the graphs of $|R|$, $|R'|$, $|R^*|$, and $|R^{*'}$ are varying significantly by a minor change in the stretch λ , as shown in Figs. (3-6)(a, b, & c). In general the maximum values of $|R|$, $|R'|$, $|R^*|$, and $|R^{*'}$ increase as stretch increases.

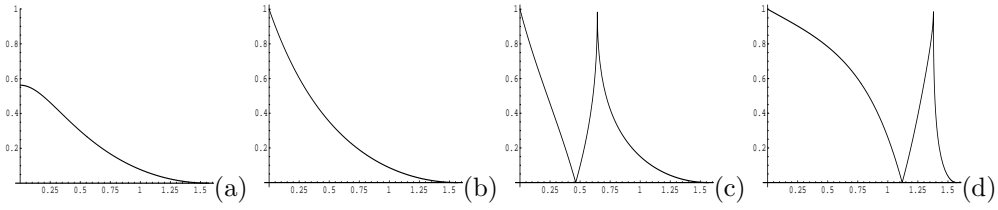


Figure 3: Plots of $|R|$ (in $x_2 < 0$) against θ ($0 \leq \theta \leq \pi/2$) with the following values of λ : (a) 1.4, (b) $\sqrt{2}$, (c) 1.5, (d) 2.5.

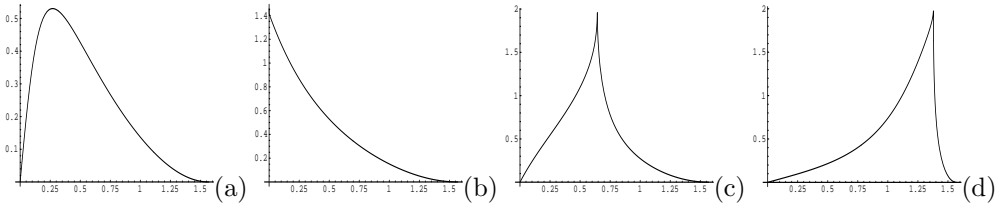


Figure 4: Plots of $|R'|$ (in $x_2 < 0$) against θ ($0 \leq \theta \leq \pi/2$) with the following values of λ : (a) 1.4, (b) $\sqrt{2}$, (c) 1.5, (d) 2.5.

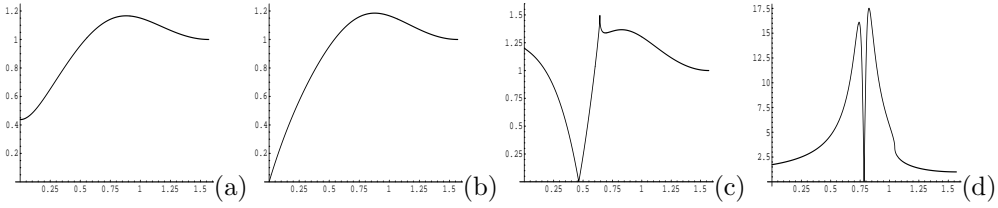


Figure 5: Plots of $|R^*|$ (in $x_2 > 0$) against θ ($0 \leq \theta \leq \pi/2$) with the following values of λ : (a) 1.4, (b) $\sqrt{2}$, (c) 1.5, (d) 1.73.

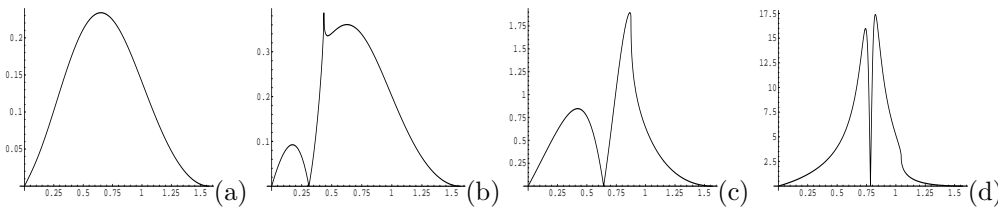


Figure 6: Plots of $|R^{*'}|$ (in $x_2 > 0$) against θ ($0 \leq \theta \leq \pi/2$) with the following values of λ : (a) 1.4, (b) 1.45, (c) 1.6, (d) 1.73.

References

- [1] W. Hussain and R. W. Ogden, Reflection and transmission of plane waves at a shear-twin interface, *Int. J. Engng Sci.*, 38 (2000), 1789–1810.
- [2] W. Hussain and R. W. Ogden, On the reflection of plane waves at the boundary of an elastic half-space subject to simple shear, *Int. J. Engng Sci.*, 37 (1999), 1549–1576.
- [3] R. W. Ogden, *Non-linear Elastic Deformations*, Dover Publications, Inc. Mineola, New York, 1997.
- [4] S. Wolfram, *Mathematica*, version 5, Wolfram Research, Champaign, Illinois, 2003.